SUPPLEMENTAL TYPE CERTIFICATE

10088573 REV. 1

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

SCHULZ, ALEXANDER

GOTTFRIED-MENKEN-STR. 23
28201 BREMEN
GERMANY

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: SEE EASA APPROVED MODEL LIST
Type Certificate Holder: SEE EASA APPROVED MODEL LIST
Type: SEE EASA APPROVED MODEL LIST
Model: SEE EASA APPROVED MODEL LIST

Description of Design Change:
Installation of electronic flight instrument on glider and motorgliders

EASA Certification Basis:
The Type and OSD Certification Bases for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.

Associated Technical Documentation:
Airplane Flight Manual supplement: Electronic Airspeed indicator + Altimeter LXNAV ADI 2, Issue 2, 14.06.2026
Supplemental type certificate (STC): Installation of a LXNAV ADI 2 to substitute Altimeter and Airspeed indicator and OAT ind., Installation Instruction, Issue 2, 14.08.2026

Or later revisions of the above listed document(s) approved/accepted under the EASA system.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 19 October 2026

Stefan FISCHER
Section Manager
Business Jets

Task Number: 60097091
SCHULZ, ALEXANDER - 311956

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Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

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